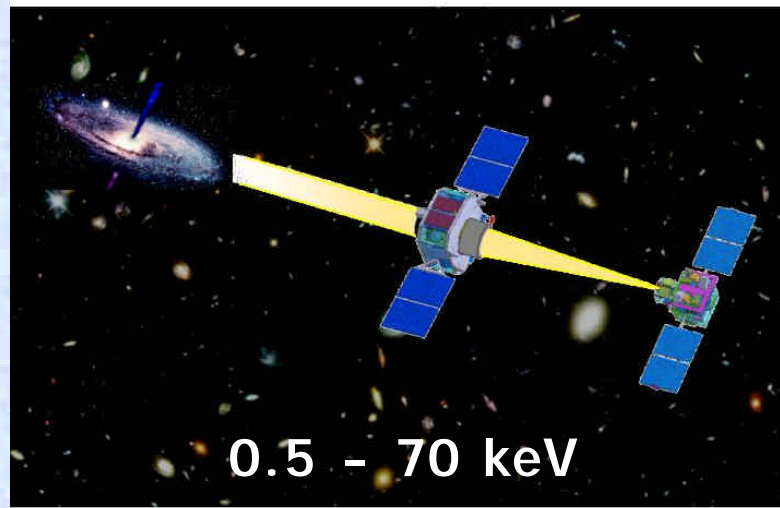


0.1-10 keV



0.5 - 70 keV



15 keV-10 MeV

Simbol-X

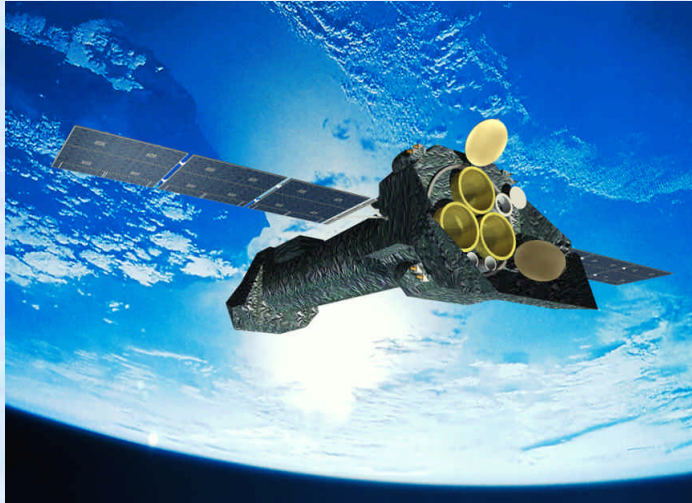
A formation flying mission for hard X-ray astrophysics

Giovanni Pareschi & Philippe Ferrando

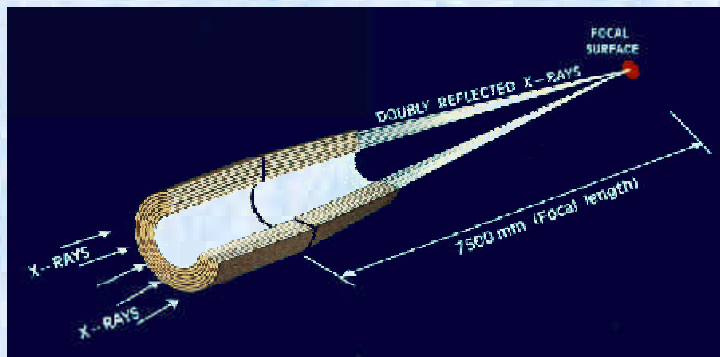
(on behalf of a large collaboration of French, Italian & German researchers)

The 10 keV sensitivity gap reason

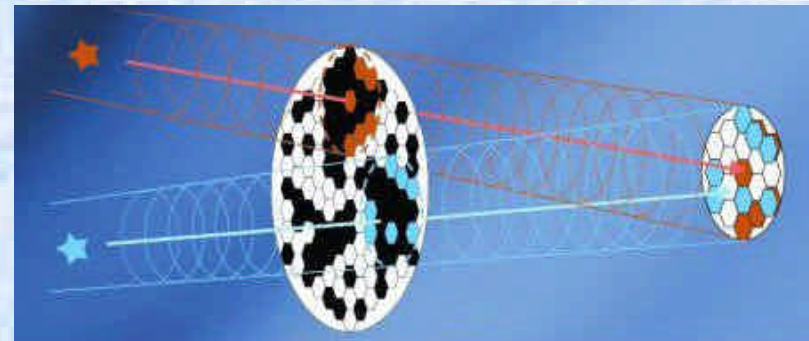
XMM-Newton



INTEGRAL

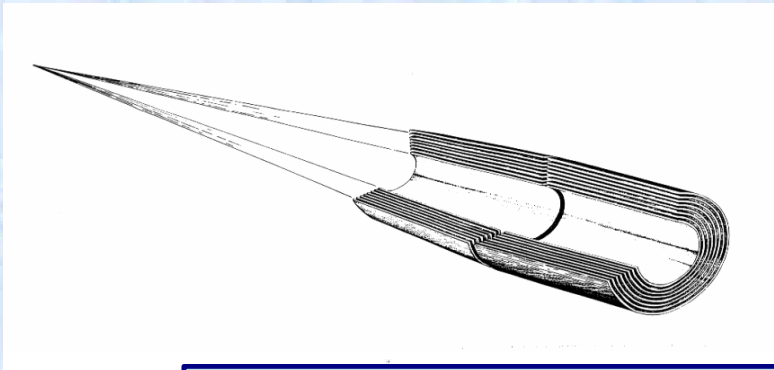


0.1–10 keV : focusing optics
Spatial resolution : 15 arcsec
High signal to noise



15 keV–10 MeV : coded masks
Spatial resolution : 12 arcmin
Moderate signal to noise

The focusing problem in the hard X-ray region (> 10 keV)



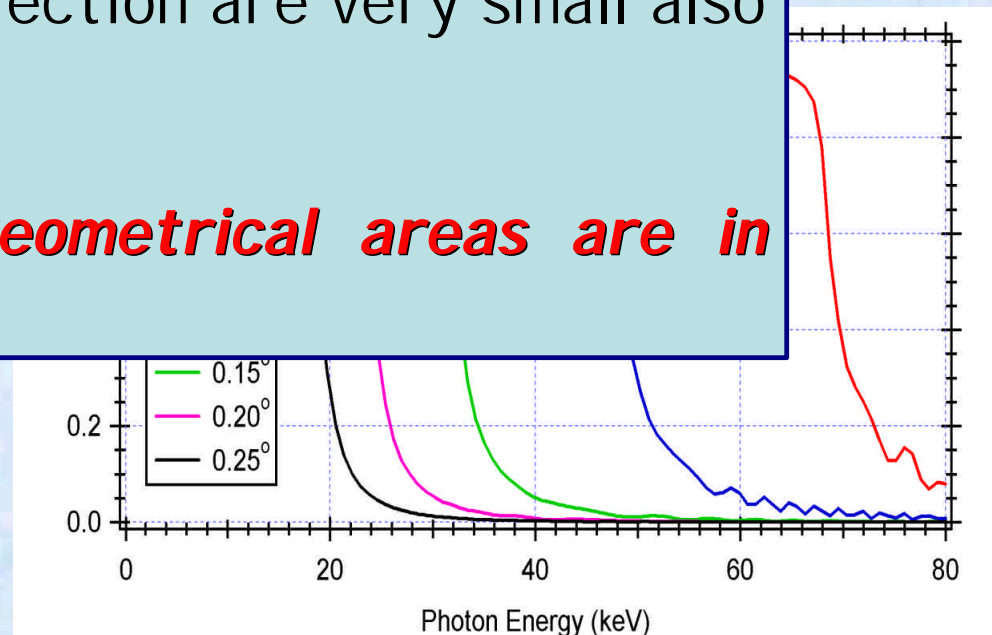
$$A_{\text{eff}} \gg F^2 \times q_c^2 \times R^2$$

$F = \text{focal length}$
 $L = \text{mirror length}$

At photon energies > 10 keV the cut-off angles for total reflection are very small also for heavy metals

→ **the attained geometrical areas are in general negligible**

L



SIMBOL-X telescope: driving design criteria

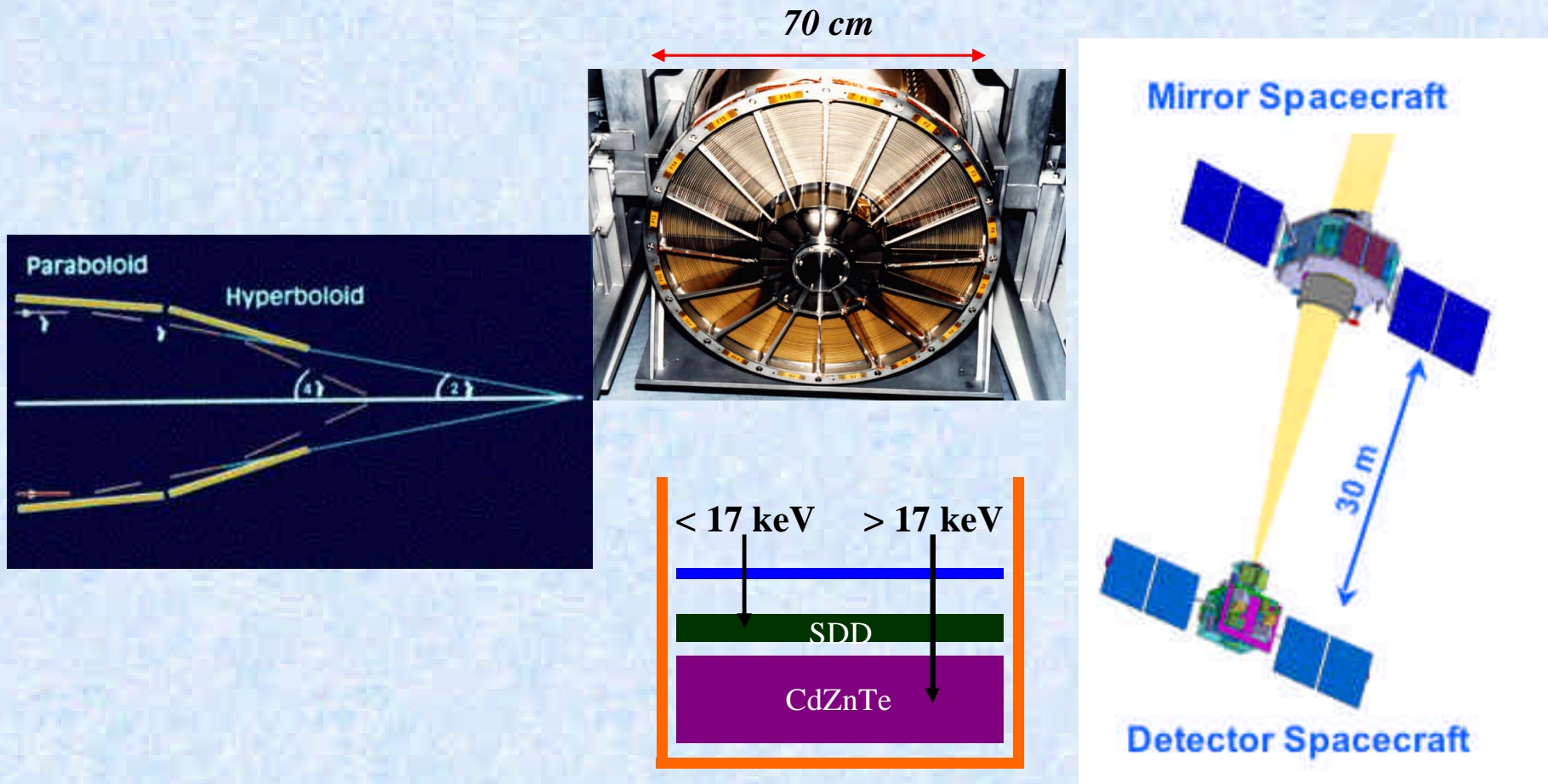
- use of high density materials coatings or multilayer mirrors (→ *increase of the reflection angles*)
- possibility to use the Ni replication technology for ***monolithic*** optics (→ already available and consolidated technology)
- maximum diameter compatible with the standard superpolishing techniques already used for XMM
- *low reflecting angles* but, at the same time, large collecting areas → ***long focal length***



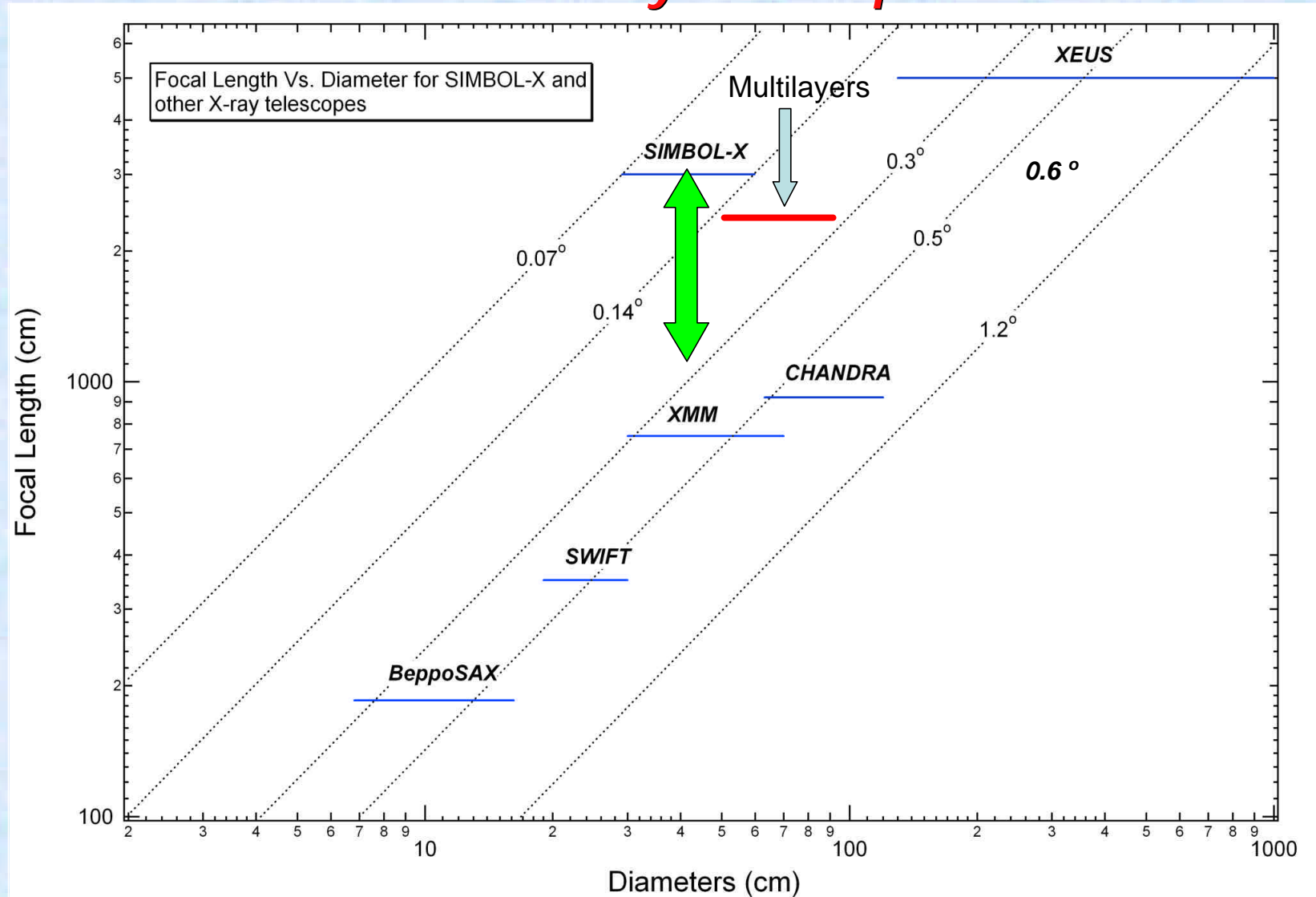
the Formation Flight architecture allow us to implement this concept!

The way to go : extend focusing into hard X-rays domain

Simbol-X proposal, basically : long focal length telescope, using grazing incidence X-ray optics, with mirror and detectors mounted on two different spacecraft in formation flying

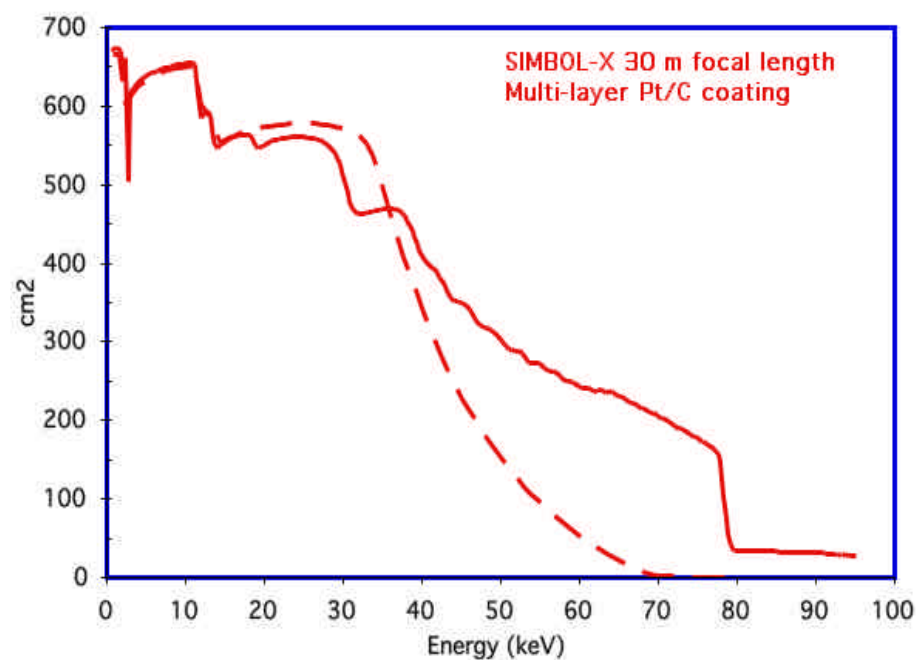
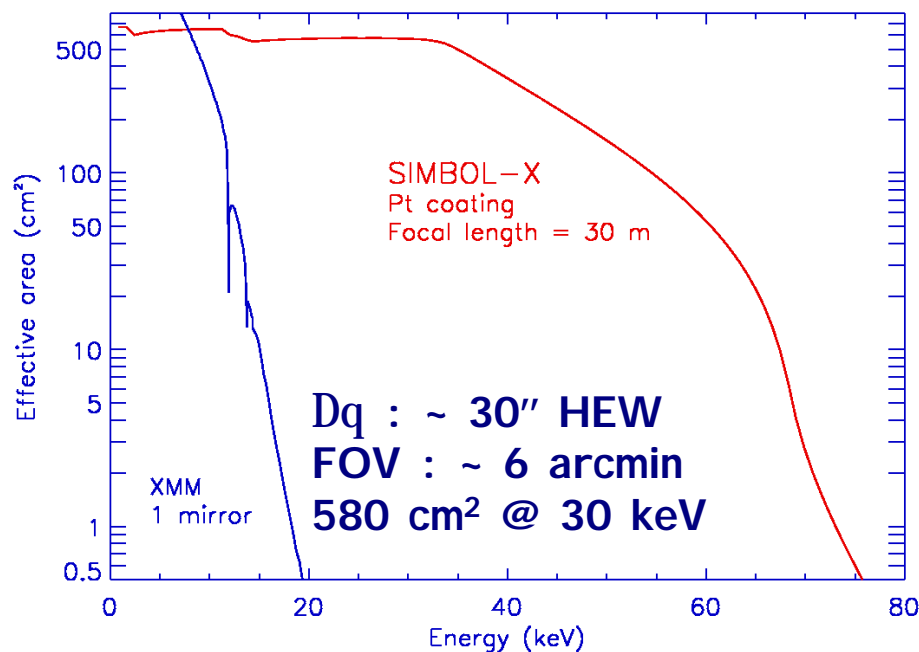


Focal Length Vs. Diameters for SIMBOL-X and other X-ray telescopes



Optics & coating

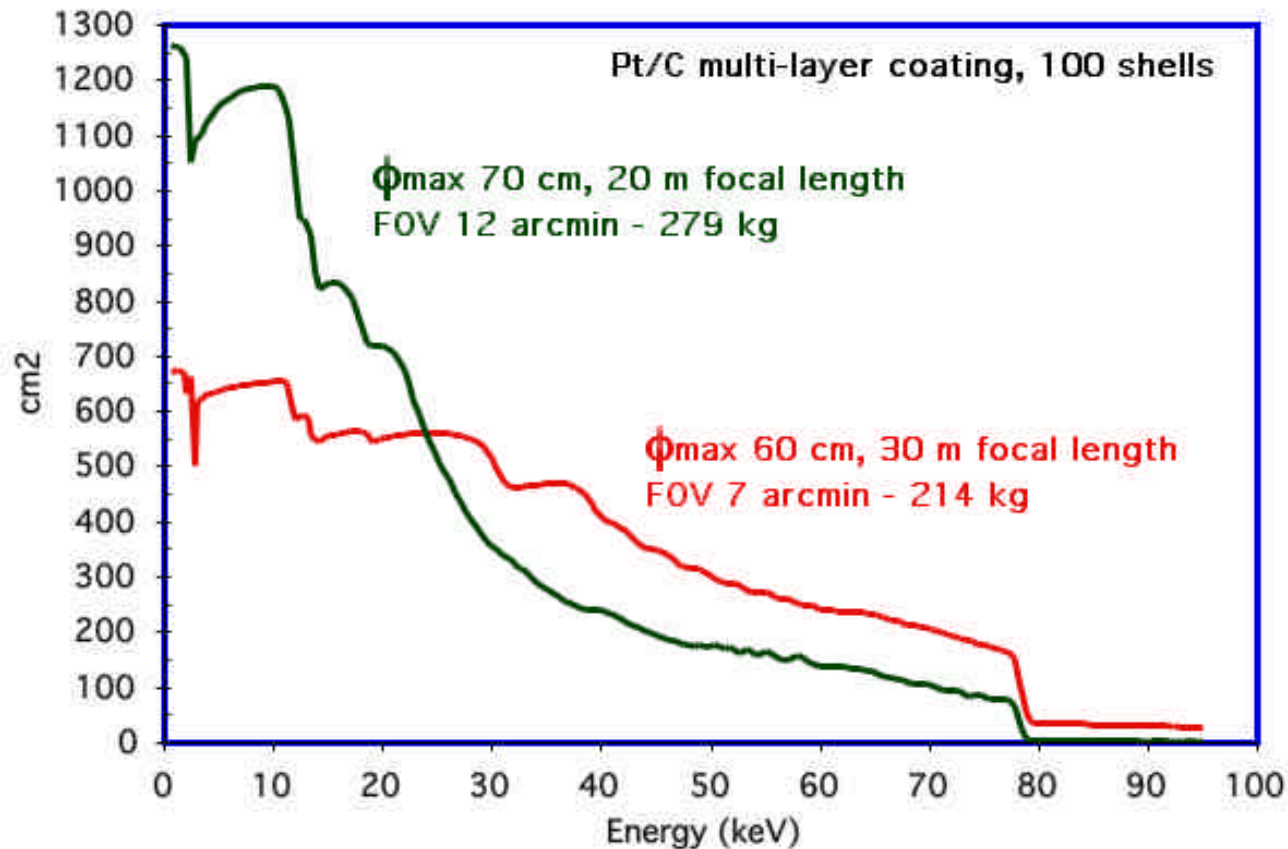
- Heritage from XMM-Newton : nickel shells obtained by electroforming replication method; low mass obtained via a reduced thickness of shells



Shell diameters : 290 to 600 mm
Angles : 0.07° to 0.142°
Shell thickness : 0.12 to 0.30 mm
Number of shells : 100
Pt coating
Total mass : 213 kg

Same parameters except for coating,
Pt/C multi layer
Strong increase of response above 40 keV
Small increase of field of view : 7 arcmin

Optics, shorter focal length option



- Strong increase of FOV and gain in plate scale, but less effective area above 25 keV, compensated by a smaller focal spot (→ less background counts)
- Trade-off to be performed between FOV, effective area, and mission constraints during phase A

Simbol-X “nominal” characteristics and potential improvements

Pt coating

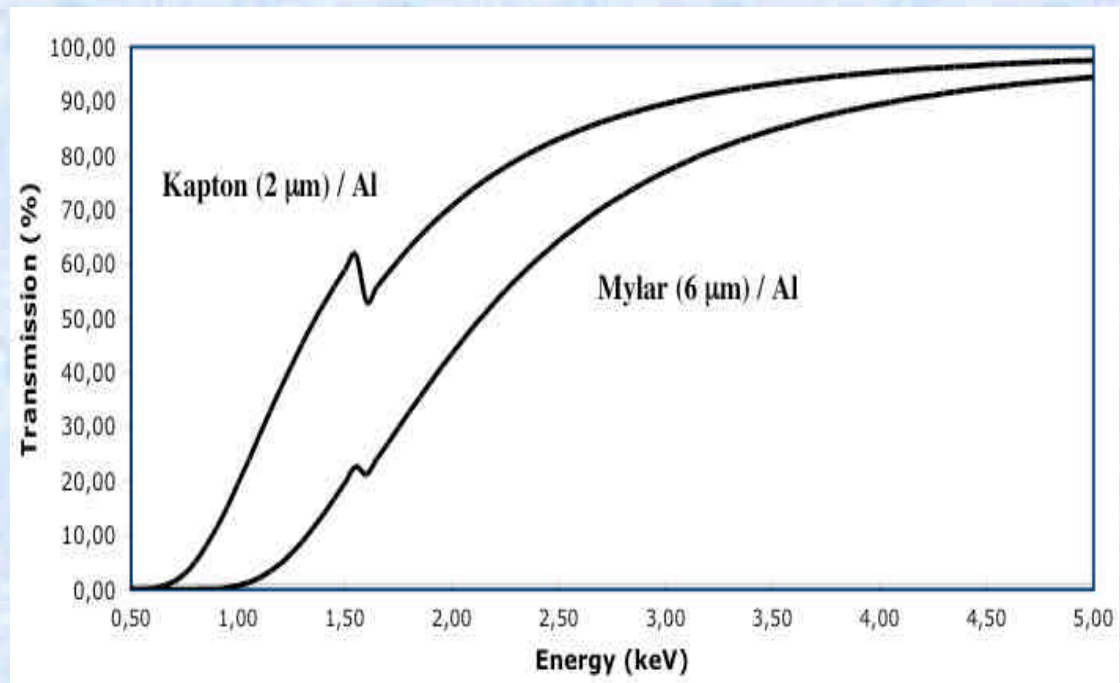
Energy range : 0.5–70 keV
Resolution : ~ 130 eV @ 6 keV
 ~ 2 keV @ 60 keV
Angular resol. : < 30 arcsec (local. < 3 arcsec)
 goal 15 arcsec
Field of View : 6 arcmin
Effective area : > 550 cm² E < 35 keV
 150 cm² @ 50 keV
 2 cm² @ 70 keV
Sensitivity : 10⁻⁸ ph/cm²/s/keV (E < 40 keV)
 (3 s, 1 Ms, DE = E/2)

Multi-layer

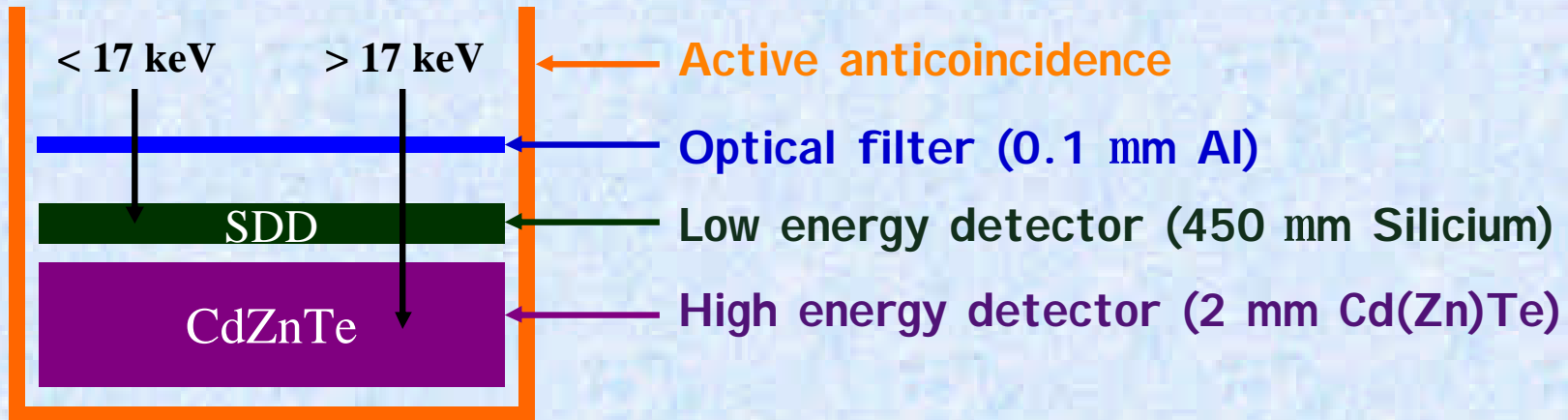
0.5 - 80 keV
1 keV @ 60 keV
7 - 12 arcmin
200 cm² @ 70 keV
up to ~ 80 keV

Mirror thermal control and low energy limit

- If mirror fully open to space (2 sides), then a very strong thermal control is needed, with ~ 200 W of heating power, and potential difficulties to ensure necessary homogeneity
- Thermal problem is much relaxed (down to ~ 20 W) by closing mirror apertures with a single foil (on each side) of insulating material
- But cost : loss of very low energies

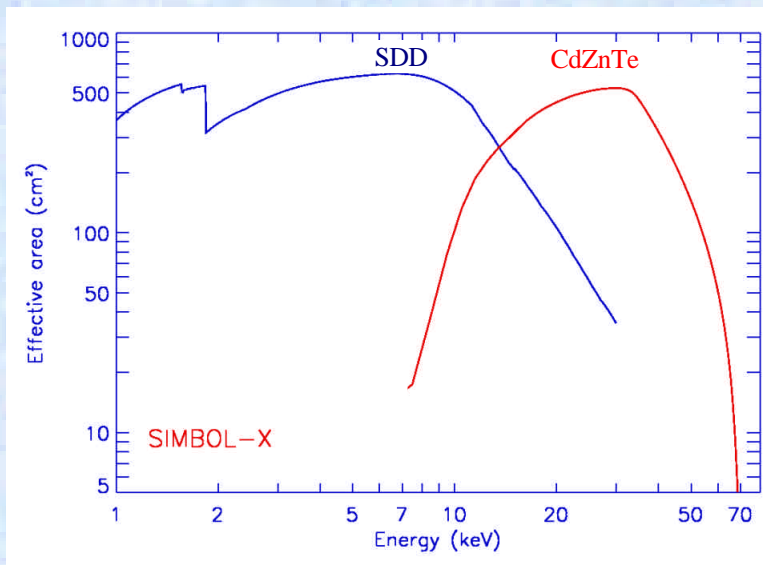


Focal plane



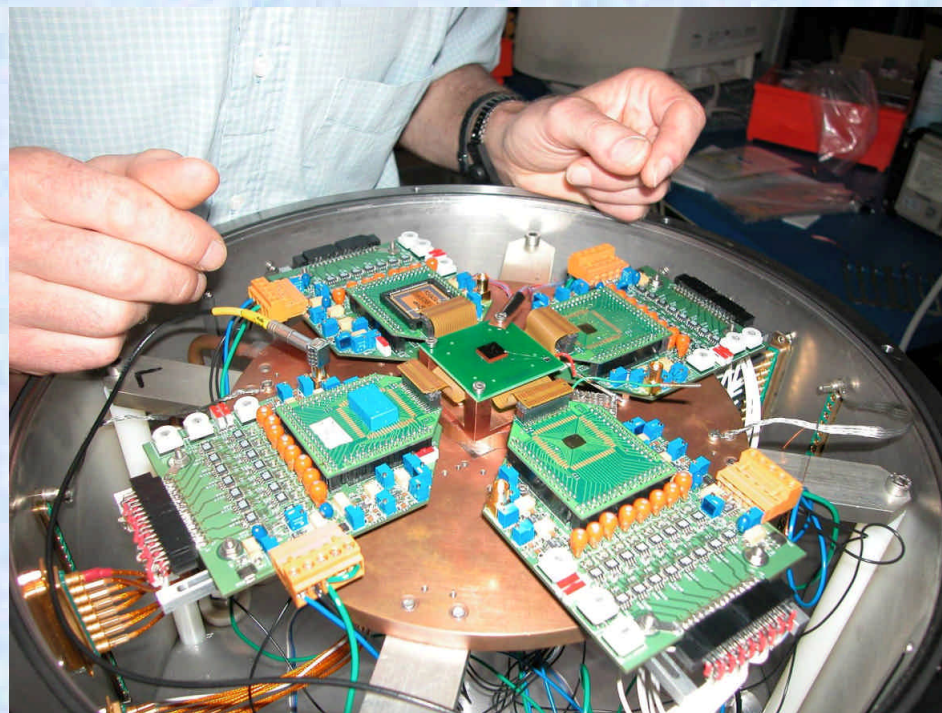
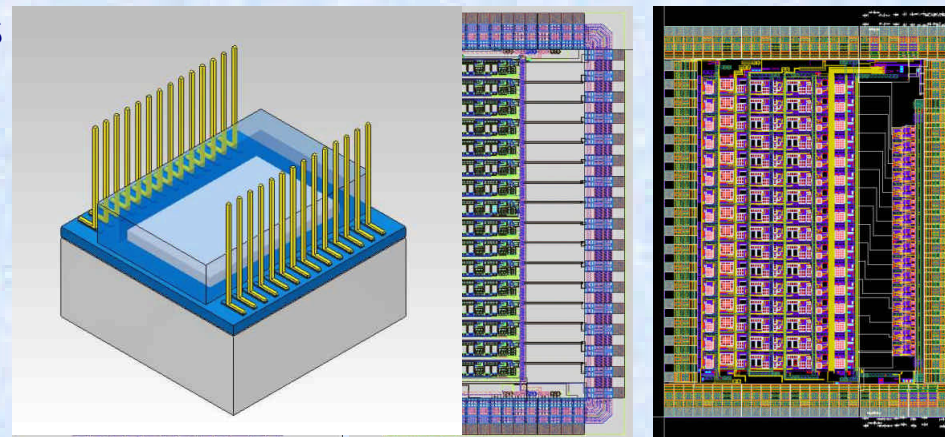
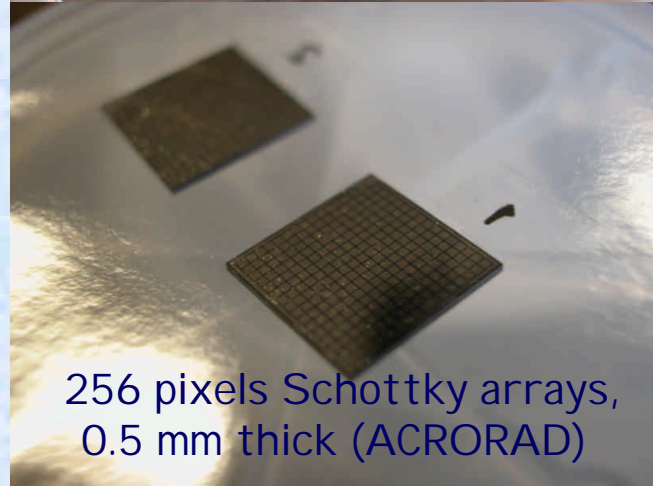
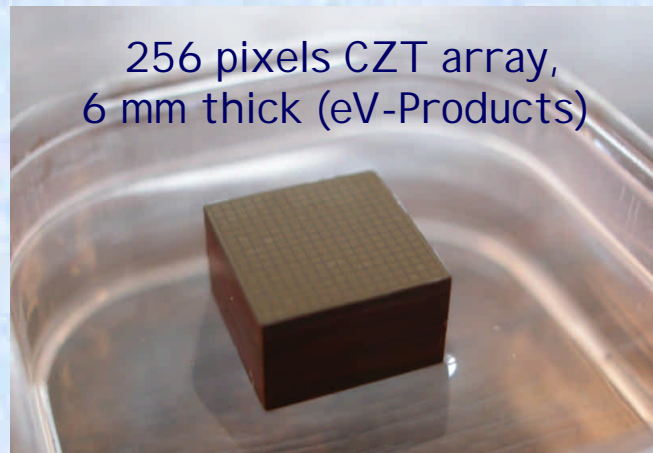
Requirements

- Pixel size of ~ 500 μm (gives good oversampling of the few mm PSF)
- Full diameter of focal plane : 6 cm
- Fast response detectors for full anticoincidence scheme
- Operation at "room" temperature
- Low energy response down to ~ 1 keV
- Good spectral resolution for Iron line

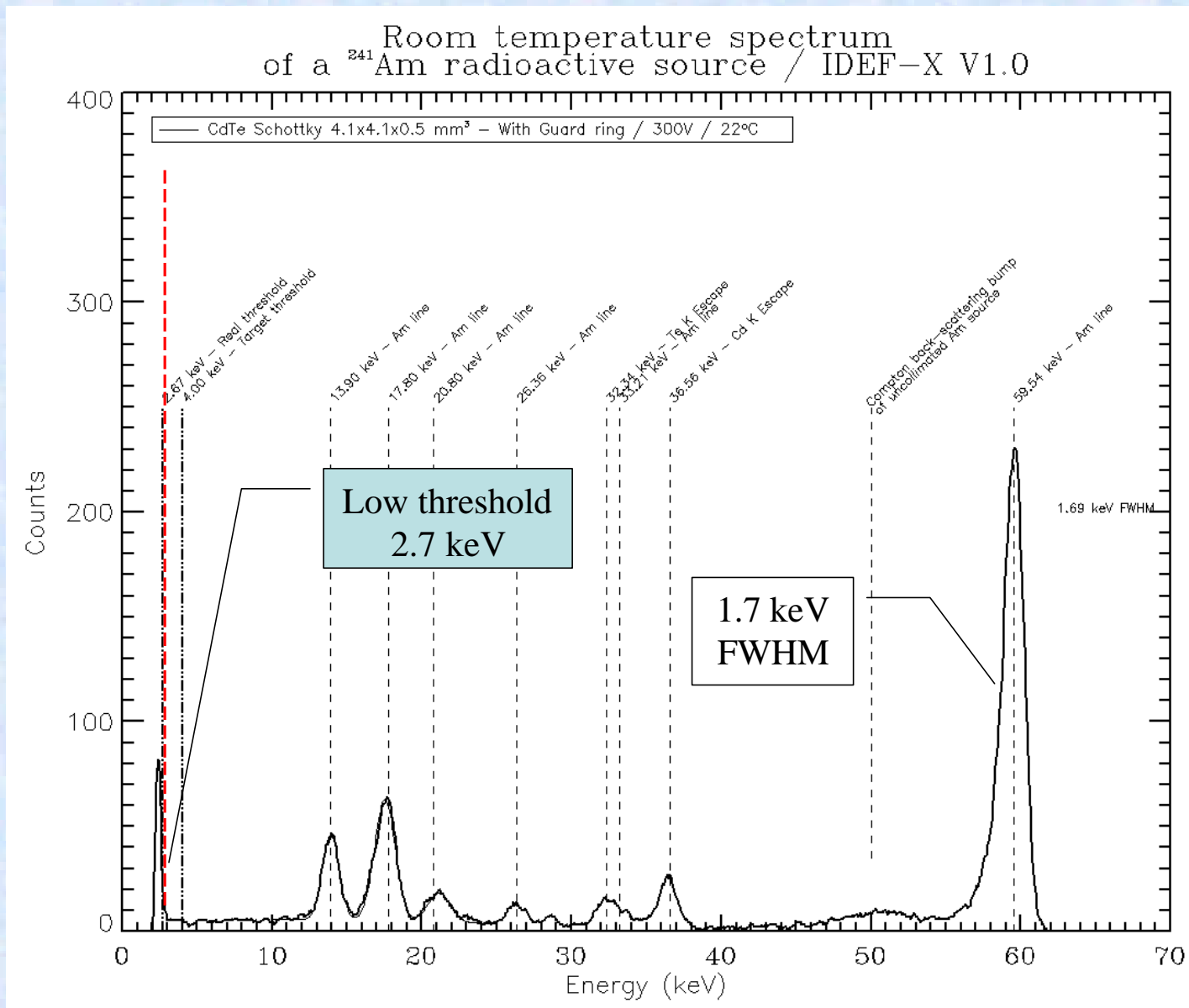


High Energy Detector : R&D in CEA (CNES funded)

- Tests pixellated Cd(Zn)Te matrices
- ASICs development (IDeF-X Vx.x)
- Hybridization

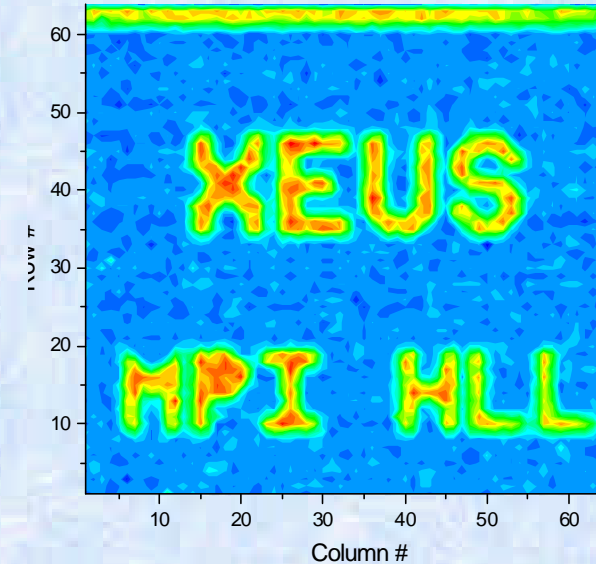
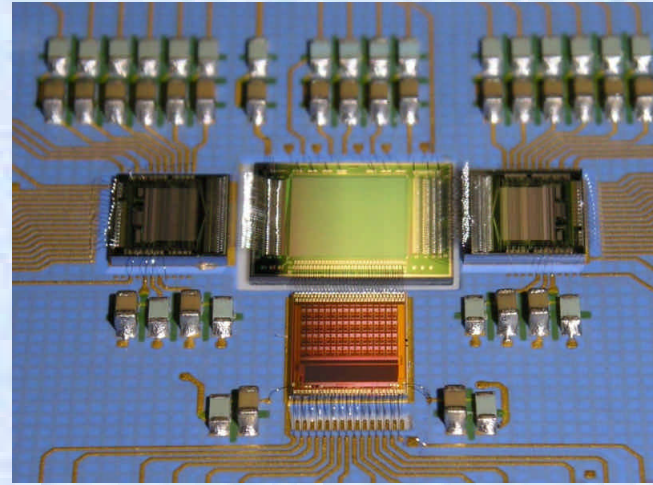
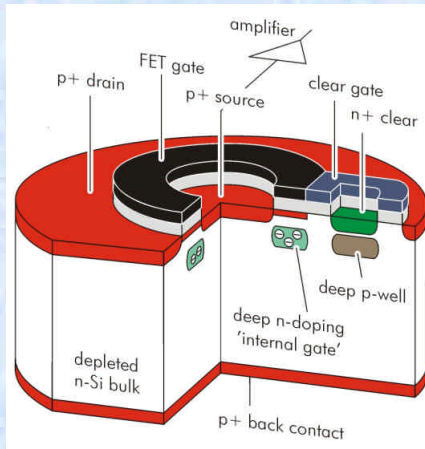


R&D in CEA, first light of IDeF-X V1.0

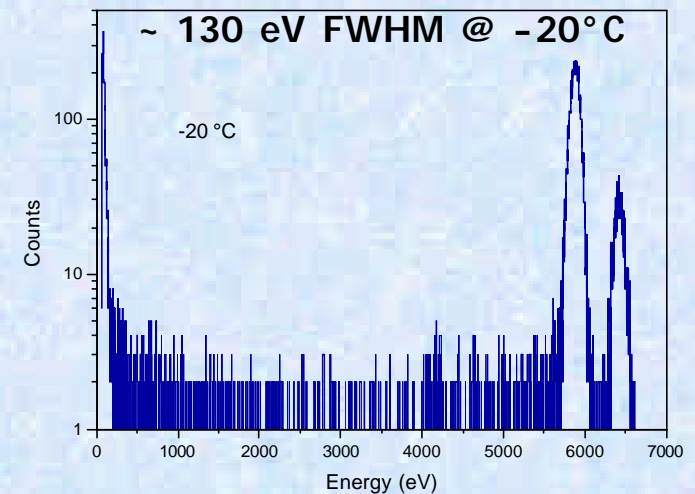


Low energy detector

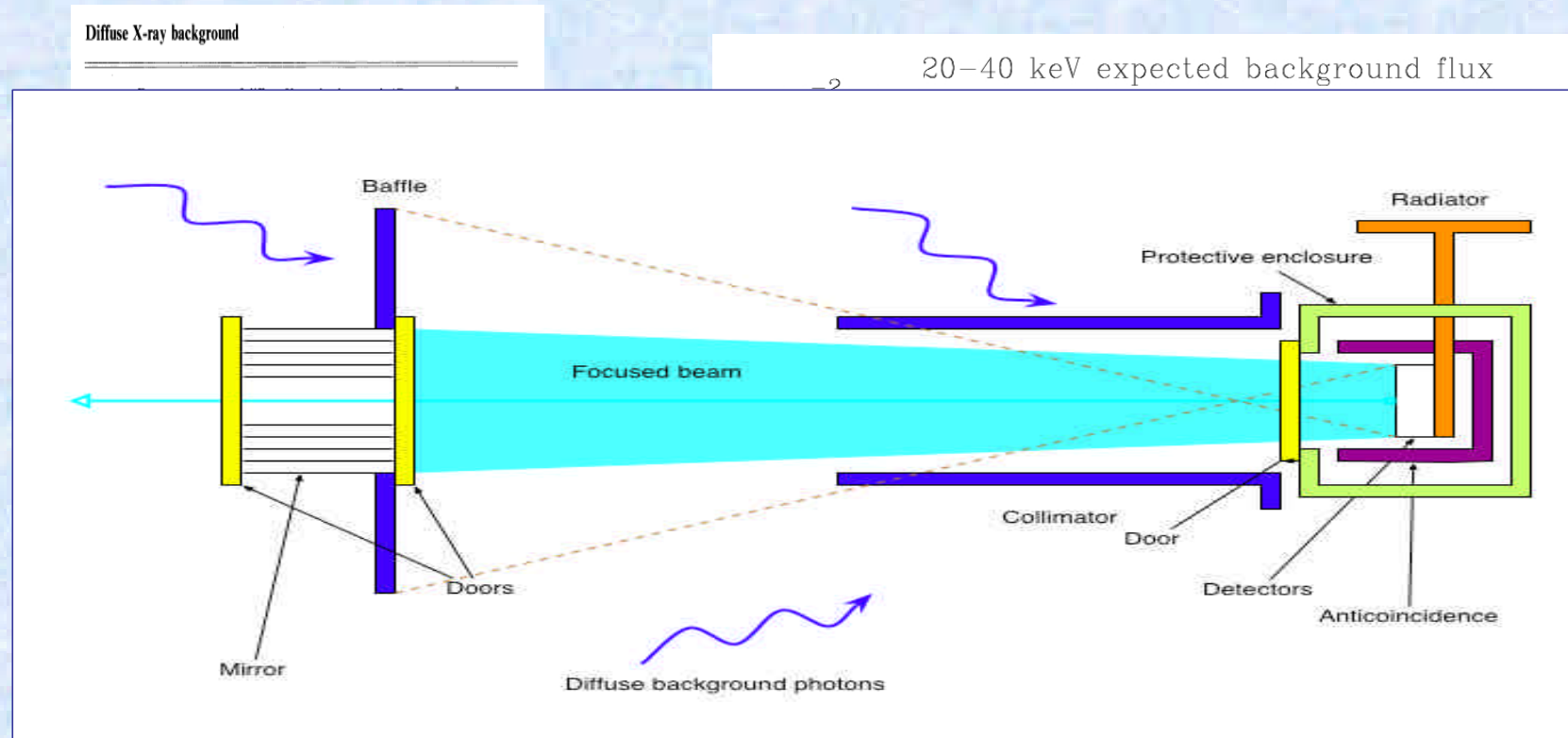
Silicon Drift Detector matrix with integrated DEPFET



- Low power consumption
- Internal amplification
- Room temperature operations
- Active Pixel Sensor type
- 100 % filling factor
- Adjustable pixel size (50 μm to 1 mm)
- Fast, parallel readout possible

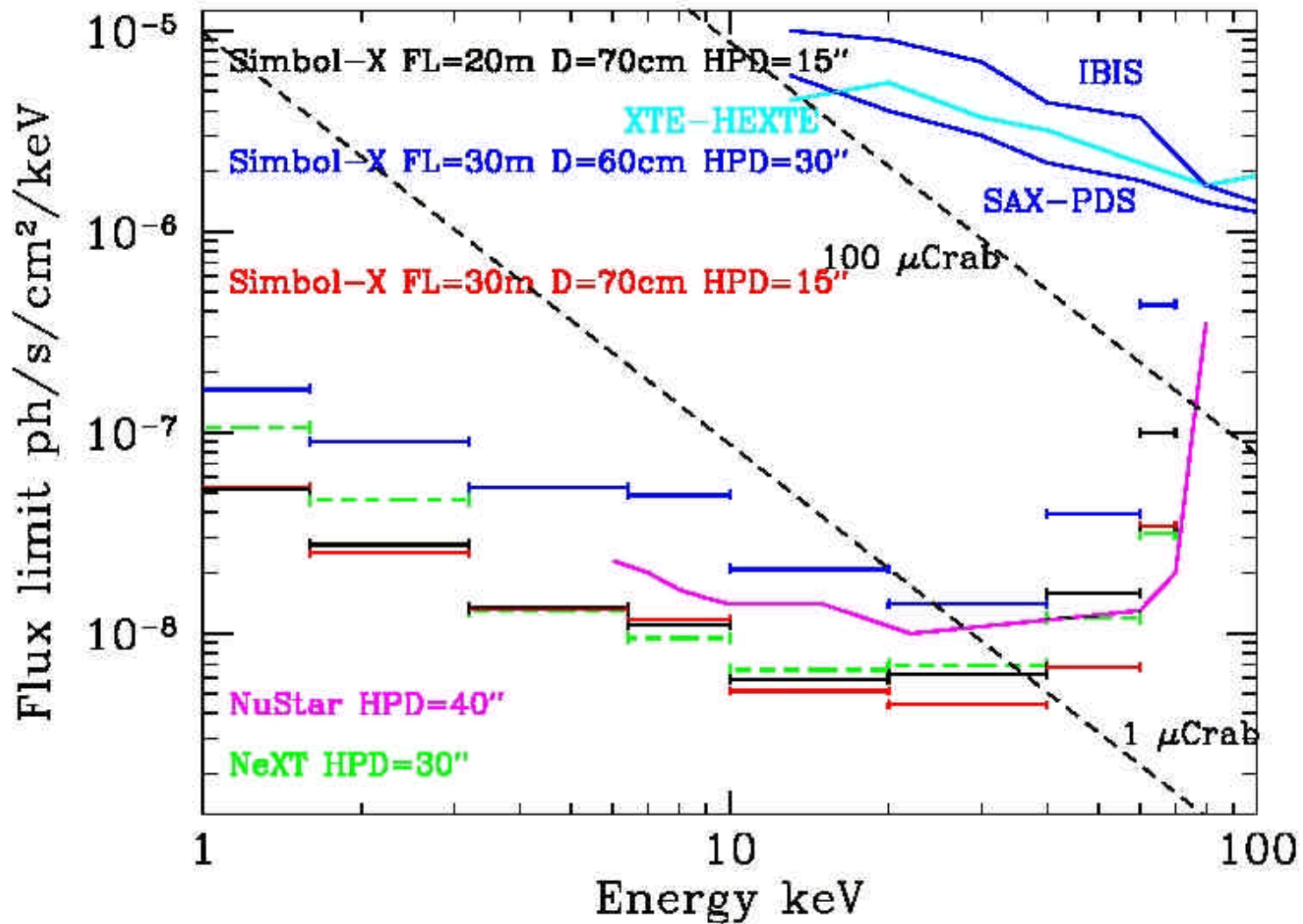


Collimator & sky-shield optimization

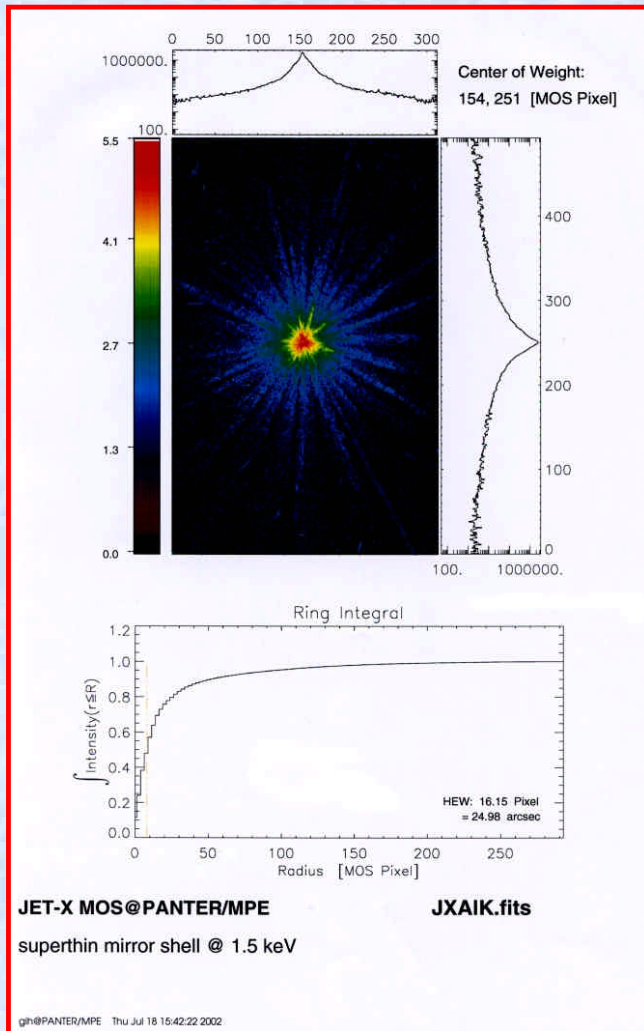


*To obtain $\sim 1 \cdot 10^{-4}$ cts $cm^{-2} s^{-1} keV^{-1}$
the collimator FOV must be $< 3^\circ$*

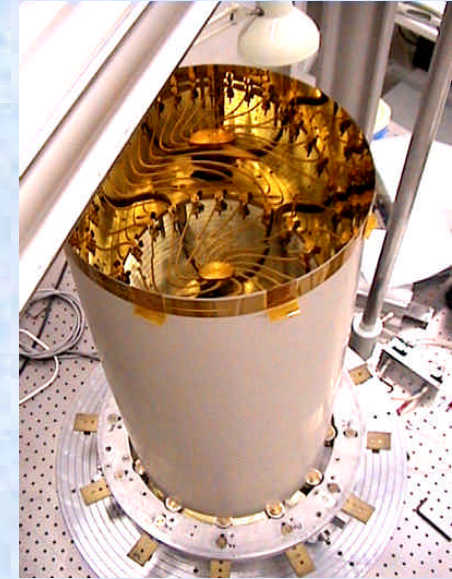
Flux sensitivity



X-ray imaging test of a thin JET-X mirror shell



- diam. = 30 cm
- thickness = 130 μm
- wall thickness **8.5 times** less than JET-X and **3.4 times** less than XMM



$$HEW_{meas} = 25 \text{ arcsec}$$

X-Ray test @ Panter-MPE
(July '02) - **E = 1.5 keV**

Simbol-X : conclusion and next phases

- Results of assessment phase : Simbol-X is mature, and well in the 30 % required margin in this study phase (total mass 2.2 tons with all system and sub-system margins, for a Soyuz capability of 2.3 tons) → talk by R. Cledassou
- No show-stoppers identified neither for the formation flying, nor for the payload
- Selection for phase A (2 missions) in October 2005
- End of phase A, and selection of the flying mission : end 2006
- Launch date for Simbol-X : end of 2012
- Operations : 2.5-3 years, with a requirement of 2 full years of science data taking, and over 1000 targets possible

- **Expression of a strong interest of ASI for a large involvement in the mission. Discussions under finalization between CNES and ASI for doing a common phase A.**

July - September 2005: Scientific topics and requirement document prepared by representative of the French & Italian community with the endorsement of ASI & CNES → **very important to fix a common baseline for pursuing the Phase A study**

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Thanks !